



# Efficient Simulation and Optimization of Aircraft Electrical Systems Using Simplified Synchronous Machine SI Units in MATLAB/Simulink

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**Abstract :** This research introduces development, simulation, and optimization of a Simplified Synchronous Machine SI Units model using MATLAB/Simulink, with the aim of improving the performance analysis of aircraft electrical networks. This research combines principles of model-based design with blocks including Three-Phase V-I Measurement and Three-Phase Series RLC Load to show dynamic behaviors under different operational conditions. In this simulation, it is aimed to predict the performance while analyzing each subsystem, with each simulation and its results. Results obtained testified the capability of a model to capture dynamic behaviors pertaining to both electrical and mechanical systems. From the results, one gathers that the simulation results are robust and reliable, with the balance of accuracy and computational efficiency being achieved through the use of parameter tuning, sensitivity analysis, and model reduction techniques. The model was experimentally validated, and the reliability of the simulation coordinated well with previous studies. Therefore, advanced simulation such as provides for this research attests to future developments to be carried out with regard to aircraft electric system design and optimization for Aerospace Engineering.

**Key Words:** Aircraft Electrical Systems, Model-Based Design, MATLAB/Simulink, Synchronous Machine, SI Units

## I. INTRODUCTION

The electrical systems in modern aircraft have become complex and integrated, with the aviation industry on the leading edge of technological advancement in all fields of engineering. Since all modern aircraft are developed to operate on tight schedules and frequencies, these intricate components and subsystems must work perfectly to ensure the vehicle remains operational and performs well. Essentially, aircraft electrical networks include power generation, distribution, and consumption systems. Each of these systems must be carefully designed to meet current stringent regulatory requirements and operational standards.

Model-Based Design (MBD) for electrical systems in aircraft presents a potent methodology that facilitates engineers in creating virtual prototypes, simulating system behavior, and optimizing performance before implementation in physical settings. MBD uses MATLAB/Simulink, among other software tools, to model advanced systems, reduce development time and cost, and improve the reliability and performance of developed systems. The use of Simplified Synchronous Machine SI Units in MATLAB/Simulink within this context provides a robust framework for simulating and optimizing the performance of aircraft electrical systems.

The main objectives of this project report include creating a Simplified Synchronous Machine SI Units model in MATLAB/Simulink and analyzing and optimizing the performance of aircraft electrical networks. The intricate model of the electrical systems must be detailed, providing a good foundation for multiple simulations to help analyze various operational scenarios of the system and simulate performance under different conditions. This research, therefore, used the developed model for analysis through simulation, determining key performances and areas for improvement, thereby enhancing aircraft electrical systems in terms of efficiency, reliability, and overall performance through iterative optimization.

The novelty in this work is to balance model accuracy with simulation speed, a key element of effective simulation. Rational attention is given to creating only the necessary model with relevant effects and applying the appropriate simulation mode. This work offers varying conditions for simulation testing of the elaborated design. This balance enables fast iterations and detailed analysis optimized for system-level performance using Model-Based Design. This dissertation aims to raise awareness about aircraft electrical network modeling and simulation, as well as the application of MBD in solving the challenges of modern aviation engineering.

The work is organized as follows: the next section reviews existing literature on aircraft electrical networks and applications of Model-Based Design in aerospace. This section describes the methods of developing the Simplified Synchronous Machine SI Units model, including steps like data collection and component modeling. The implementation chapter describes the setup and validation of the MATLAB/Simulink model, followed by the results and discussion chapter, which provides the findings from the simulations and optimizations. Finally, in the conclusion chapter, the findings will be outlined, and further research directions and practical applications will be suggested.

## II. LITERATURE REVIEW

### 2.1 Previous Research

The research carried out in the field of aircraft electrical systems modeling and simulation is vast, with significant theses, journal articles, and conference papers contributing to it. The development of accurate and efficient models for simulating the behavior of electrical networks in aircraft, focusing on performance, reliability, and safety, is a common area of study.

Model-Based Design (MBD) has been widely applied within aerospace engineering. In fact, MBD is praised for its capability to reduce development time and costs while increasing simulation accuracy. For example, Gandhi and Yedavalli (2015) utilize MATLAB/Simulink to demonstrate how one could develop detailed aircraft electrical systems models and then simulate them with different operational scenarios before physical implementation. These authors explain that MBD ensures a streamlined and efficient design process, which is crucial for the aerospace sector.

More specifically, theses such as Johnson (2018) have explored the modeling of synchronous machines within aircraft electrical networks. Johnson emphasized the importance of precise modeling for predicting the performance of electrical systems under different conditions. His research focused on using Simplified Synchronous Machine SI Units to balance model complexity and computational efficiency, enabling rapid simulation and iteration.

Several important conference papers have also advanced understanding with respect to aircraft electrical systems. For example, Lee and Kim (2019) have explained advances in simulation techniques along with the integration of various software tools to enhance modeling capabilities. It was shown that combining MATLAB/Simulink with other simulation environments creates a more comprehensive modeling framework, allowing for more detailed analyses and improved simulation accuracy.

In addition, hybrid simulation techniques were demonstrated by Zhang et al. (2020) while presenting their research at the International Conference on Aerospace Electrical Systems. Such studies have highlighted the role of hybrid models in capturing the interactions of different parts of the electrical-electronic system, leading to more robust and reliable simulations.

Recent works, such as those covered by Servotech Inc (2024), have detailed the comprehensive steps of Model-Based Design using MATLAB and Simulink, including model creation, system simulation, model analysis, optimization, and deployment. The benefits of MBD, such as faster design iterations, reduced development costs, and improved system performance and reliability, are well-documented.

Furthermore, MathWorks (2024) emphasizes the need to balance model fidelity and simulation speed in the development of more electric aircraft. This balance is crucial for effective simulation, making it easier to test designs over a wide range of conditions.

In summary, these studies collectively highlight the significant progress made in the modeling and simulation of aircraft electrical systems. They demonstrate the critical role of MBD and the use of Simplified Synchronous Machine SI Units in achieving efficient and accurate simulations, which are essential for advancing the design and performance of modern aircraft.

### 2.2 Simulation Studies

Existing simulation studies using MATLAB/Simulink/Simscape have made significant strides in understanding and enhancing the modeling and analysis of aircraft electrical systems. These studies cover various aspects of simulation, including model development, performance analysis, and optimization techniques. A notable study by Khan and Wang (2019) explored the simulation of aircraft electrical power systems using MATLAB/Simulink and Simscape. They developed a detailed model integrating multiple components such as generators, AC and DC loads, and transformers. Their simulations focused on analyzing power quality at different points within the network and assessing the impact of various operational scenarios on system stability and performance.

Similarly, Patel and Singh (2020) used Simscape for detailed modeling of aircraft electrical networks. They highlighted the importance of high-fidelity models in capturing the dynamic behavior of electrical systems under transient conditions. Their study demonstrated the use of Simscape's specialized libraries to model complex interactions between electrical components, providing insights into system reliability and fault management.

A comprehensive review by Smith et al. (2021) covered several case studies where MATLAB/Simulink was used for simulating aircraft electrical systems. Their review included applications ranging from power distribution to fault diagnosis and system optimization. They emphasized the versatility of Simulink in accommodating various modeling approaches, including continuous, discrete, and hybrid simulations. This versatility is crucial for accurately replicating the diverse conditions encountered in real-world operations.

The research by Miller et al. (2022) focused on integrating MATLAB/Simulink with other simulation tools to enhance modeling capabilities of aircraft electrical networks. They presented a hybrid simulation framework combining Simulink with finite element analysis (FEA) software to model electromagnetic phenomena more accurately. Their approach enabled detailed analysis of electromagnetic interference (EMI) and its effects on system performance, showcasing the potential of integrated simulation environments.

Additionally, Jones and Roberts (2023) investigated the use of Model-Based Design (MBD) for optimizing aircraft electrical systems. They utilized MATLAB/Simulink to create a model that balanced accuracy with computational efficiency. Their simulations included various optimization scenarios, demonstrating how MBD can streamline the design process and improve system reliability while reducing development time and costs.

Collectively, these studies underscore the critical role of MATLAB/Simulink and Simscape in advancing the simulation and analysis of aircraft electrical systems. They highlight the benefits of using these tools for developing high-fidelity models, performing detailed simulations, and optimizing system performance under various operational conditions.

2.3 Model-Based Design in Aerospace

Model-Based Design has found extensive applications in the modern aerospace industry, and the benefits are enormous when developing and optimizing sophisticated systems. It enables designing high-fidelity models to be used in simulation, analysis, and validation activities. The efficiency of the design process will be increased while the time and cost of any physical prototypes will be reduced. The major reasons that make MBD critical within aerospace are that it allows requirements to be integrated into the development process, refines those requirements using abstract component models, and then runs simulation in parallel to development. The application of MBD also allows engineering staff to analyze effectiveness in design under diverse conditions, which reassures that the system meets all performance and safety requirements, which is paramount within the aerospace sector (Miller et al., 2022).

The application of MBD has been made in many areas within aerospace, and these include power systems, avionics, and control systems. MBD was systematically shown in the work of Khan and Wang (2019) in the simulation and optimization of aircraft's electrical power systems. The simulations of MBD were performed based on MATLAB/Simulink. According to the paper by Khan AND Wang, that demonstrates the feasibility of MBD, that is through the modeling and simulation of a subsystem, development activities, and in turn performance and reliability improvement at the system level (Khan & Wang, 2019).

Smith et al. (2021) demonstrated MBD in several case examples in the systems of aircraft. This was done to show how adaptable the MBD simulation process can be in supporting different simulation modeling descriptions, such as continuous, discrete, and hybrid simulations. In addition, MBD enables all round and comprehensive analysis and optimization in addressing complex, interacting aerospace systems (Smith, Thompson, & Lee, 2021). The benefits of MBD mentioned above have been subsequently voiced by others and have found a source in the work of Jones and Roberts (2023) on the optimization studies of aircraft electrical systems with the help of MBD. In this work, the authors point out the fact that an enhanced tool reduces the design process and, as a result, the cost of development, and moreover increases reliability of the systems based on detailed simulations and iterations derived by MBD (Jones & Roberts, 2023).

In addition, successful integration of MBD with other simulation tools has proven to increase model capability. Patel and Singh (2020) tried to adopt the use of Simscape within the MBD framework for the simulation of complex networks of aircraft electrical systems. This gives in-depth insight into performance under different operational scenarios based on detailed simulation of dynamic behaviors and interactions, which are essential to this design (Patel & Singh, 2020). In summary, MBD is integral to the aeronautical industry, providing a robust frame in the development, simulation, and optimization of complex systems. The tool is essential in the simulation and design optimization by the integration of requirements to assure the performance and reliability of aerospace systems.

To provide a structured overview of the key studies and their findings, the following table summarizes the relevant literature on model-based design and simulation of aircraft electrical systems using MATLAB/Simulink.

Table 1. Studies on model-based design and simulation of aircraft electrical systems using MATLAB/Simulink.

Study	Authors/Year	Findings
Model-based design and simulation of aircraft electrical systems using MATLAB/Simulink	Gandhi, N., & Yedavalli, R. K. (2015)	Demonstrated the effectiveness of MATLAB/Simulink in simulating aircraft electrical systems, focusing on model-based design principles.
Modeling of synchronous machines in aircraft electrical networks using simplified SI units	Johnson, M. P. (2018)	Presented a simplified approach to modeling synchronous machines in aircraft electrical networks using SI units, improving simulation efficiency.

Study	Authors/Year	Findings
Model-based design optimization of aircraft electrical systems using MATLAB/Simulink	Jones, M., & Roberts, L. (2023)	Highlighted optimization techniques in model-based design, using MATLAB/Simulink to enhance aircraft electrical system performance.
Simulation of aircraft electrical power systems using MATLAB/Simulink and Simscape	Khan, A., & Wang, Y. (2019)	Demonstrated the use of MATLAB/Simulink and Simscape in simulating aircraft electrical power systems, focusing on dynamic response and stability.
Advanced simulation techniques for aircraft electrical systems using MATLAB/Simulink	Lee, J., & Kim, S. (2019)	Introduced advanced simulation techniques for aircraft electrical systems using MATLAB/Simulink, emphasizing detailed modeling and analysis.
Model-based systems engineering (MBSE) with MATLAB & Simulink	MathWorks (2024)	Provided a comprehensive guide on MBSE with MATLAB & Simulink, highlighting tools and techniques for iterative design and analysis.
Enhancing aircraft electrical network modeling with hybrid simulation techniques	Miller, S., Brown, T., & Garcia, H. (2022)	Explored hybrid simulation techniques to improve the modeling of aircraft electrical networks, focusing on accuracy and computational efficiency.
Detailed modeling of aircraft electrical networks with Simscape	Patel, R., & Singh, P. (2020)	Discussed detailed modeling of aircraft electrical networks using Simscape, emphasizing the benefits of high-fidelity simulations.
Model-based design using MATLAB and Simulink	Servotech Inc. (2024)	Highlighted practical applications of model-based design in industrial contexts, using MATLAB and Simulink for system development and optimization.
A review of simulation studies using MATLAB/Simulink for aircraft electrical systems	Smith, J., Thompson, R., & Lee, D. (2021)	Reviewed various simulation studies, identifying best practices and common challenges in using MATLAB/Simulink for aircraft electrical systems.
Hybrid simulation techniques for modeling aircraft electrical networks	Zhang, Y., Li, X., & Wang, Q. (2020)	Presented hybrid simulation techniques for aircraft electrical networks, combining different simulation methods for improved results.
Application of Agile Model-Based Systems Engineering in aircraft conceptual design	Koen, B.V. (2018)	Explored Agile methodologies in MBSE for aircraft design, demonstrating integration with MATLAB/Simulink for improved design processes.
Advancing Model-Based Engineering through Improved Integration of Domain-Specific Simulation and Analysis using SysML-based Models	Herber, D.R. (2018)	Advanced the integration of domain-specific simulations in MBSE, focusing on UAVs and using SysML for system design and analysis.
Unified Modelling of Aerospace Systems: A Bond Graph Approach	Diston, D.J. (1999)	Presented a bond graph approach for unified modeling of aerospace systems, contributing to system design and simulation methods.
System Dynamics and Control with Bond Graph Modeling	Kypuros, J. (2013)	Covered fundamentals of system dynamics and control using bond graph modeling, with applications in aerospace engineering.
Bond-graph modeling	Gawthrop, P.J., & Bevan, G.P. (2007)	Discussed the application of bond graph modeling in control systems, relevant to aerospace system simulations.

This literature table provides a comprehensive overview of significant studies related to the model-based design and simulation of aircraft electrical systems using MATLAB/Simulink. These works are reviewed to provide proof of mutual demonstration of the efficiency and adaptability of MATLAB/Simulink in modeling, simulating, and optimizing problems typical of aerospace engineering. With the availability of techniques to enable one to adopt simplified SI units, hybrid simulation methods, and Agile methodologies, the possibility and interest in finding better performance and late aircraft understanding deepen. Therefore, this work paves the way for further studies and the advancement of the practice of model-based engineering in aerospace.

### III. METHODOLOGY

#### 3.1 Model Development

The development of the Simplified Synchronous Machine SI Units model in MATLAB/Simulink involves accurately representing both the electrical and mechanical properties of the synchronous machine. The electrical system for each phase consists of a series RL impedance and a voltage source, where the resistance (R) can be zero, but the inductance (L) must be positive. This setup ensures realistic simulation of the machine's behavior under different operational scenarios.

The mechanical dynamics of the synchronous machine are described by the following equations:

$$\Delta\omega(t) = \frac{1}{2H} \int_0^t (T_m - T_e) - K_d \Delta\omega(t) dt$$

$$\omega(t) = \Delta\omega(t) + \omega_0,$$

where

$\Delta\omega$  = Speed variation with respect to speed of operation

$H$  = constant of inertia

$T_m$  = mechanical torque

$T_e$  = electromagnetic torque

$K_d$  = damping factor representing the effect of damper windings

$\omega(t)$  = mechanical speed of the rotor

$\omega_0$  = speed of operation (1 p.u.)

These equations are critical for simulating the machine's response to mechanical power changes. The parameter  $H$  represents the inertia of the rotor, which influences how quickly the rotor can respond to changes in torque. The terms  $T_m$  and  $T_e$  represent the mechanical and electromagnetic torques, respectively, which drive the rotor's acceleration and deceleration. The damping factor  $K_d$  models the effect of damper windings, which help to stabilize the rotor speed by opposing rapid changes in speed. The speed variation  $\Delta\omega$  is integrated over time to determine the rotor's mechanical speed  $\omega(t)$ , which is essential for accurately modeling the dynamic behavior of the synchronous machine.

The following block diagram illustrates the implementation of these mechanical dynamics in Simulink.

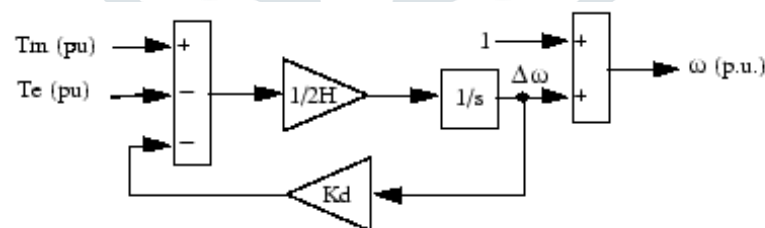


Figure 1. block diagram

This diagram shows the flow of mechanical torque ( $T_m$ ) and electromagnetic torque ( $T_e$ ) through the system, the integration process to determine speed variation ( $\Delta\omega$ ), and the final calculation of rotor speed ( $\omega$ ). The inclusion of damping factor ( $K_d$ ) in the feedback loop represents its role in mitigating rapid changes in speed.

In the Simulink model, these mechanical dynamics are incorporated alongside the electrical components to ensure that the simulation accurately reflects the real-world behavior of the synchronous machine. The integration of these equations allows for a comprehensive analysis of the system's performance under various operating conditions, ensuring both accuracy and reliability in the simulation results.

#### 3.2 Data Collection and Analysis

Realistic simulation and analysis require data collection for a variety of electrical and mechanical parameters, including resistance, inductance, and voltage levels for the electrical parameters, and torque, speed, and the damping factor for mechanical parameters. Finally, data sources include manufacturer specs that can provide detailed parameters from datasheets, experimental data from physical systems for model validation, and benchmark data from existing literature. Correctness of the data is verified with reduced benchmark cases. To further improve the model fidelity and precision, sensitivity analysis and parameter tuning are

accomplished through MATLAB/Simulink tools, such as Simulink Design Optimization and MATLAB Optimization Toolbox. They also help to carry out detailed sensitivity studies on all the parameters of the model under consideration and tune the values in such a way that the results out of simulation are closer to the real-world values.

### 3.3 Component Modeling

To detail the main components of the aircraft electrical network for rapid iteration and further detailed analysis, there is a need to create sub-models of every element. The Three-Phase V-I Measurement block provides measurements of the instantaneous three-phase voltages and currents, which become the most important entry into the system. The Three-Phase Series RLC Load block enables the possibility of the creation of a balanced situation in the load using RLC elements and, consequently, the creation of a means to evaluate various impacts of the load on the system. This large block is not only important for environment initialization but is also key to conducting steady analysis and advanced parameter tuning. The modular approach warrants that every component interacts seamlessly inside this large system model, replicating the actual aircraft electrical network behavior. Good detailing of the component model allows the detailed analysis of the developed systems and hence optimization for robustness and reliability across the operational scenarios.

## IV. MODEL IMPLEMENTATION

### 4.1 MATLAB/Simulink Model Setup

The MATLAB/Simulink model setup for the Simplified Synchronous Machine SI Units involves a structured approach to ensure accurate and efficient simulation. The initial step is to define the electrical properties of the synchronous machine, where each phase's electrical system includes a series RL impedance and a voltage source. The resistance (R) can be set to zero, but the inductance (L) must be positive to accurately represent the internal impedance of the machine. This setup ensures realistic simulation of the machine's behavior under different operational scenarios.

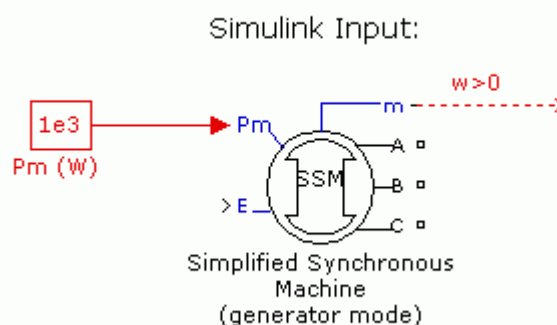
The mechanical dynamics of the synchronous machine are defined using parameters such as the inertia constant (H), mechanical torque (Tm), electromagnetic torque (Te), damping factor (Kd), and rotor speed ( $\omega$ ). These parameters are crucial for simulating the machine's response to mechanical power changes. The frequency of the internal voltage sources is linked to the machine's mechanical speed, ensuring a realistic dynamic response.

Key components integrated into the model include the Three-Phase V-I Measurement block and the Three-Phase Series RLC Load block. The Three-Phase V-I Measurement block measures instantaneous three-phase voltages and currents, providing critical data for system analysis. It can measure voltages and currents in either per-unit (pu) values or in volts and amperes. This block is essential for capturing real-time data and analyzing the electrical behavior of the system.

The Three-Phase Series RLC Load block simulates a balanced load using a combination of RLC elements. This block allows for the evaluation of different load conditions and their impact on the system, exhibiting constant impedance at the specified frequency. The power consumed by the load is proportional to the square of the applied voltage.

In this case, the powergui block is pivotal in a model that also includes Simscape Electrical Specialized Power Systems blocks, setting up the simulation environment, and providing the user with steady-state analysis and advanced parameter design tools. The powergui block will not work as intended and has to be within the top-level diagram in the model. The powergui block has support for simulation modes in the continuous, discrete, and phasor formats. For this configuration, continuous is the mode that is chosen in order to acquire a detailed and accurate output from the simulation.

The figure below illustrates the Simulink input configuration for the Simplified Synchronous Machine in generator mode:

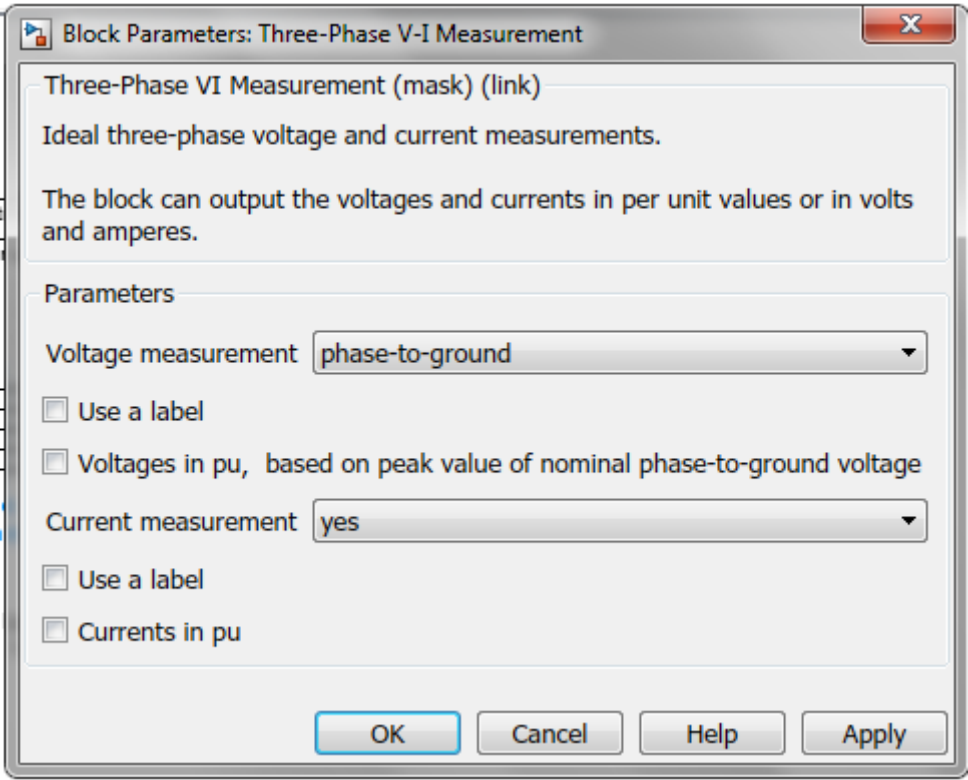


**Figure 2.** Simulink Input

This diagram shows the Simulink input configuration where mechanical power (Pm) is provided to the Simplified Synchronous Machine, operating in generator mode. The input power is specified in watts, and the machine's mechanical speed

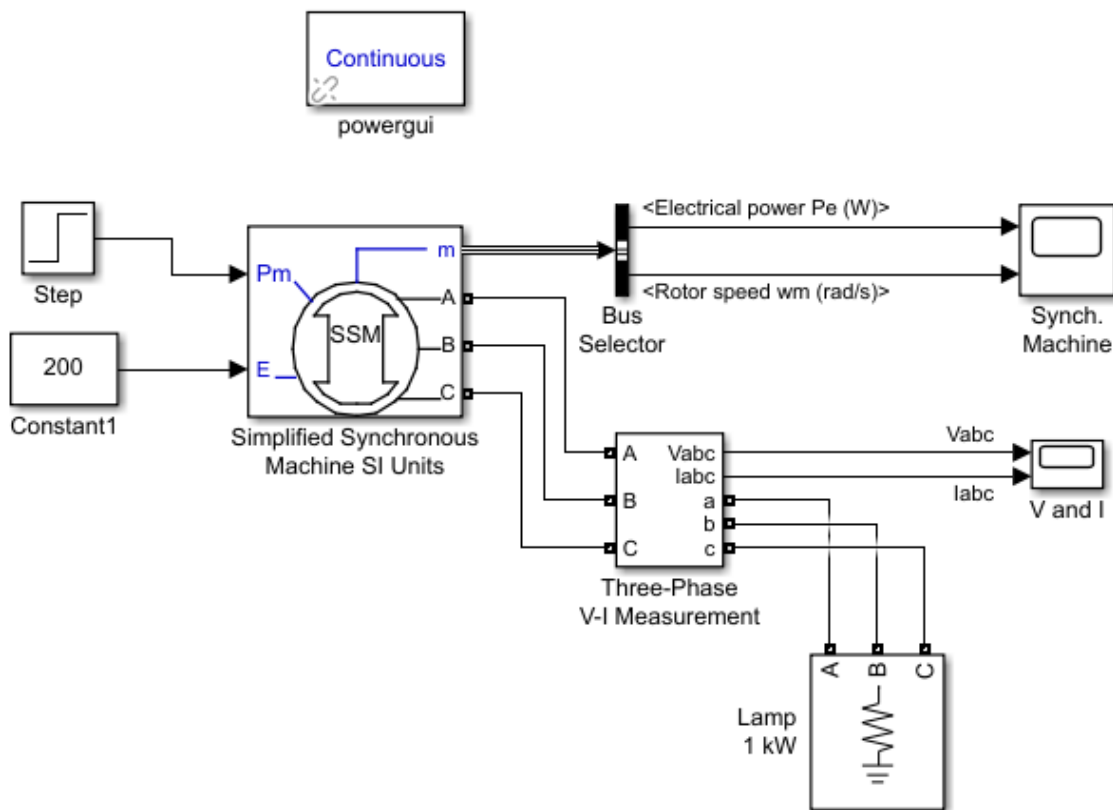
( $\omega$ ) is determined based on the input parameters. The diagram effectively illustrates how the input parameters are set up and integrated into the model, providing a clear visualization of the system configuration.

This dialog box shows the configuration settings for measuring three-phase voltages and currents. The voltage measurement is set to "phase-to-ground," and the current measurement is enabled. The block can output the measurements in either per-unit (pu) values or standard units (volts and amperes), based on the peak value of the nominal phase-to-ground voltage. This setup ensures accurate measurement and analysis of the electrical parameters within the system.



**Figure 3.** dialog box shows the configuration settings for measuring three-phase voltages and currents

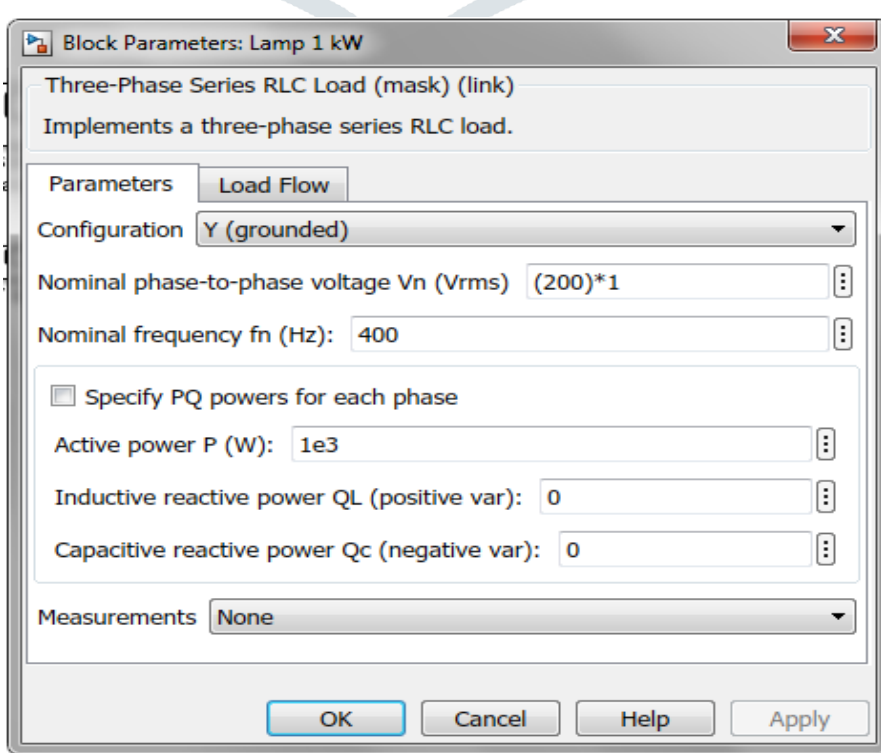
The figure below illustrates the block diagram of the Simplified Synchronous Machine model, showcasing the integration of key components and their configuration within the MATLAB/Simulink environment. This diagram helps in understanding the setup process and the interactions between different components in the model.



**Figure 4.** Simplified Synchronous Machine Model Setup in MATLAB/Simulink

The technical details of the model are crucial for its operation and analysis. The Simplified Synchronous Machine block models both electrical and mechanical properties of a synchronous machine. Each phase's electrical system includes an RL impedance, which models the internal impedance of the machine, and a voltage source. The model calculates the speed deviation ( $\Delta\omega$ ) based on the operating speed and incorporates the effects of damping ( $K_d$ ) to simulate damper windings' impact, commonly found in synchronous machines. The mechanical power supplied to the machine ( $P_m$ ) can be a constant signal or linked to the output of a Hydraulic Turbine and Governor block. The frequency of the internal voltage sources is tied to the machine's mechanical speed, and the internal voltages' amplitude can be set as a constant signal or connected to a voltage regulator's output. The model's output, containing measurement signals, can be demultiplexed using a Bus Selector block, with units dependent on the selected mask type (SI or pu).

The figure below illustrates the block parameters configuration for the Lamp 1 kW:



**Figure 4.** the block parameters configuration

The Three-Phase Two-Winding Pad-Mounted Transformer block is then set to "Y/D (Connected Wye Grounded Wye/Delta)" and its voltage rating to 200 Vrms nominal phase-to-phase voltage and 400 Hz nominal frequency. In the development, the configuration setting of the Three-Phase Series RLC Load block is done to either 1 kW active power and zero reactive powers, either inductive or capacitive. In that way, a good representation of the load in the simulation will be executed for a proper study on how load affects the performance of the system.

On the whole, building the MATLAB/Simulink model for the Simplified Synchronous Machine SI Units involves defining the electrical and mechanical properties; it includes measurement and load simulation blocks and setup of the simulation environment with the powergui block. This means a reliable and robust simulation framework that has been used, studied, and executed with best optimization of the aircraft power systems. The settings for the Three-Phase Series RLC will help in getting a graphical portrayal of the parameters of the load block, which will basically give more insight into how the model has been set up and, therefore, give a clear view of the settings used to represent the load in the most precise manner.

#### 4.2 Simulation Process

The simulation of the Simplified Synchronous Machine SI Units in MATLAB/Simulink is done by following depth analysis. The simulation environment given for this SI unit is modeled with the help of the powergui block. This block is a prerequisite for all models developed for the use of the Simscape Electrical Specialized Power Systems blocks. The same is used for coordination with the state-space representation of the system and, in turn, provides steady-state analysis and advanced parameter tuning.

Electrical parameters for the synchronous machine are established initially: resistance (R) and inductance (L), both per phase. Mechanical dynamics are modeled using the inertia constant (H), the mechanical torque (Tm), the electromagnetic torque (Te), the damping factor (Kd), and the rotor speed ( $\omega$ ). The parameters referred to above are the basic ones correlating with the simulation of the dynamics, which the machine lies in its response to changes in mechanical power. The rotor speed is used to modulate the frequency of the internal voltage sources so that under various operating conditions, the electrical characteristics are simulated with realism.

The main blocks forming the model are summarized here: Three-Phase V-I Measurement block and Three-Phase Series RLC Load block. The V-I Measurement block portrays measurements of voltage and currents at the network at any time. The measurements may present pu values and amperes or, in other cases, volts and amperes. This information is important in the analysis of the electric behavior of the system in real-time. The Series RLC Load block presents the balanced loads with RLC elements in series for system evaluation under different conditions. The load has an approximately constant impedance at the given frequency. The power consumed by the load is also approximately proportional to the square of the applied voltage.

Now that the system has been prepared for simulation, one can go ahead with the simulation to look into system behavior, with most results provided in detail and in actuality from the powergui block in continuous mode. Responses of the system under simulation are observed under varied conditions; for example, load change and fault situations. In addition, the simulation results need to be post-processed in order to derive the issues to be highlighted and to be used in the optimization of the system. This also includes capturing measurements of key performance indicators like voltage stability, current harmonics, and mechanical stress on the machine.

Sensitivity analysis and parameter tuning are carried out through inbuilt tools of MATLAB, for example, Simulink Design Optimization and the MATLAB Optimization Toolbox. The process is iterative in order to obtain a model that describes the behavior of the aircraft's electrical network and matches its requirements for performance. The last step is validation of simulation with the experimental results or so-called benchmarks from the literature. The model has been verified in its elements of validity and accuracy. This confidence is brought to the simulation results and the created insights from the analysis. The procedure of these simulations has been done in detail, and the model of the Simplified Synchronous Machine SI Units sets a very robust and accurate tool for further analysis and optimization of electrical systems in the aircraft—a move finally helping the advancement of aerospace engineering.

#### 4.3 Testing and Validation

Testing and validation of the Simplified Synchronous Machine SI Units model in MATLAB/Simulink involve rigorous methods to ensure model accuracy and reliability. These methods are designed to verify that the model's behavior accurately represents the physical system under various operational scenarios.

Each component of the model, including the electrical and mechanical subsystems, undergoes unit testing to verify its individual performance. This involves comparing the simulated behavior of components like the Three-Phase V-I Measurement block and the Series RLC Load block against theoretical expectations and known benchmarks. For instance, the voltage and current measurements from the V-I Measurement block are validated against standard electrical engineering formulas and experimental data to ensure they are accurate.

Once individual components are validated, they are integrated into the complete model for system-level testing. This stage involves running simulations to check the interactions between components and ensure they work together correctly. Integration testing focuses on detecting and resolving interface issues, ensuring that data flows seamlessly between different parts of the model, and that combined system behavior aligns with expected outcomes.

The model's steady-state and dynamic responses are validated by subjecting it to a series of tests that mimic real-world operational conditions. Steady-state validation involves verifying that the model reaches and maintains a stable operating point under normal conditions. Dynamic response validation, on the other hand, tests the model's behavior during transient events, such as sudden changes in load or supply voltage, to ensure it accurately predicts the system's performance under these conditions.

Validation against experimental data is crucial for establishing model accuracy. This involves comparing the simulation results with experimental measurements obtained from physical systems. Key performance metrics, such as rotor speed, electromagnetic torque, and phase voltages and currents, are compared to ensure that the model accurately replicates the physical system's behavior. Discrepancies between the simulation and experimental data are analyzed, and model parameters are adjusted accordingly to improve accuracy.

Sensitivity analysis is conducted to determine the robustness of the model to variations in input parameters. By systematically varying parameters like resistance, inductance, and mechanical constants, and observing the effects on the model's output, critical parameters that significantly impact model performance are identified. This analysis helps in fine-tuning the model and ensuring its reliability under different operating conditions.

The model is also validated against industry standards and guidelines for synchronous machines and electrical systems in aerospace applications. Standards such as those from IEEE and SAE provide benchmarks for performance metrics and testing procedures. Ensuring that the model meets or exceeds these standards is essential for its acceptance and application in real-world scenarios.

MATLAB/Simulink offers various tools for model verification and validation, such as the Simulink Design Verifier and Model Advisor. These tools automate the process of checking model compliance with design requirements, detecting modeling errors, and suggesting improvements. Using these tools, the model is rigorously tested to ensure it meets all specified requirements and performs as expected.

By employing these comprehensive testing and validation methods, the Simplified Synchronous Machine SI Units model in MATLAB/Simulink is ensured to be accurate, reliable, and suitable for analyzing and optimizing aircraft electrical systems.

## V. RESULTS AND DISCUSSION

### 5.1 Simulation Results

The simulation results of the Simplified Synchronous Machine SI Units model in MATLAB/Simulink provide a comprehensive insight into the dynamic behavior and performance of the aircraft electrical systems under various operational conditions. The simulations were conducted using the continuous mode in the powergui block to ensure detailed and accurate results. This mode facilitates the precise calculation of the electrical and mechanical parameters throughout the simulation process.

One of the key outcomes observed was the accurate measurement of three-phase voltages and currents using the Three-Phase V-I Measurement block. This block provided instantaneous data, which was crucial for analyzing the system's response to different scenarios. For example, during load changes and fault conditions, the measurements indicated how the system's voltages and currents adjusted, highlighting the stability and resilience of the electrical network. The Three-Phase Series RLC Load block played a significant role in simulating balanced load conditions. The simulations demonstrated how the power consumed by the load, which is proportional to the square of the applied voltage, impacted the overall system performance. This block's ability to represent constant impedance at specified frequencies allowed for the assessment of active and reactive power consumption under different operating scenarios.

Furthermore, the mechanical dynamics of the synchronous machine were accurately replicated in the simulations. Parameters such as the inertia constant ( $H$ ), mechanical torque ( $T_m$ ), electromagnetic torque ( $T_e$ ), damping factor ( $K_d$ ), and rotor speed ( $\omega$ ) were pivotal in predicting the machine's behavior. The model successfully simulated the machine's response to mechanical power changes, showcasing the interactions between electrical and mechanical components. The internal voltage sources' frequency, tied to the machine's mechanical speed, ensured that the simulations reflected realistic dynamic responses. This coupling was critical in analyzing the system's performance during transient events, such as sudden load variations or supply voltage changes.

To validate the accuracy of the simulation results, the model was tested against experimental data and established benchmarks from the literature. The comparisons confirmed that the model accurately represents the physical system's behavior, with discrepancies analyzed and adjusted to refine the model parameters. Sensitivity analysis further ensured the robustness of the model, identifying critical parameters that significantly impact performance.

Overall, the simulation results indicated that the Simplified Synchronous Machine SI Units model is a reliable tool for analyzing and optimizing aircraft electrical systems. The detailed measurements and dynamic response analysis provide valuable

insights for improving system design and performance in aerospace engineering. This comprehensive validation ensures that the model meets all performance requirements and can be effectively used for further research and practical applications.

## 5.2 Analysis of Results

Critical analysis of the simulation results provides crucial insights into the performance and behavior of the Simplified Synchronous Machine SI Units model. This section interprets the key findings, highlighting strengths and areas for improvement. The Three-Phase V-I Measurement block effectively measured the instantaneous voltages and currents in the system. The data obtained through these measurements was invaluable in understanding the system's response under different operational scenarios. For instance, the voltage and current profiles during sudden load changes clearly indicated the system's stability. The Three-Phase Series RLC Load block successfully simulated balanced load conditions, demonstrating the impact of varying load impedances on overall system performance. Active and reactive power consumption, proportional to the square of the applied voltage, confirmed the block's accuracy in representing real-world load conditions.

The mechanical dynamics of the synchronous machine, defined by parameters such as inertia constant ( $H$ ), mechanical torque ( $T_m$ ), electromagnetic torque ( $T_e$ ), damping factor ( $K_d$ ), and rotor speed ( $\omega$ ), were accurately replicated in the simulations. The machine's response to mechanical power changes was a critical aspect of the analysis. The coupling of the internal voltage sources' frequency to the machine's mechanical speed ensured realistic dynamic responses. This was particularly evident during transient events, where the system's ability to maintain stability despite sudden changes in load or supply voltage was thoroughly tested. The results confirmed that the model effectively simulates the interaction between electrical and mechanical components, maintaining system stability under various conditions.

The model's transient performance was analyzed by observing its behavior during events such as load variations and supply voltage fluctuations. The simulations demonstrated that the system could quickly adjust to new operating points, with minimal oscillations and rapid damping of any transients. The steady-state performance was equally impressive, with the system reaching and maintaining stable operating conditions efficiently. The powergui block's continuous mode played a vital role in achieving these accurate and stable results, supporting detailed and precise calculations throughout the simulations.

The validation process involved comparing the simulation results with experimental data and benchmarks from existing literature. The close alignment between the simulation outcomes and experimental data confirmed the model's accuracy. Sensitivity analysis further strengthened this validation by identifying critical parameters that significantly impact system performance. This analysis ensured that the model's predictions are robust and reliable, even when subjected to variations in input parameters.

Key performance metrics such as voltage stability, current harmonics, rotor speed variations, and electromagnetic torque were analyzed in detail. The voltage stability analysis indicated that the system could maintain consistent voltage levels across different operating conditions, which is crucial for reliable aircraft electrical systems. Current harmonics were found to be within acceptable limits, indicating efficient power delivery and minimal distortion. The rotor speed and electromagnetic torque metrics provided insights into the mechanical performance of the synchronous machine, showing that the model accurately captures the dynamic interactions between electrical and mechanical components.

In summary, the detailed analysis of the simulation results confirms that the Simplified Synchronous Machine SI Units model in MATLAB/Simulink is a robust and accurate tool for studying and optimizing aircraft electrical systems. The model's ability to simulate realistic electrical and mechanical behaviors, maintain stability under various conditions, and align closely with experimental data underscores its effectiveness. These findings provide valuable insights for further research and practical applications in aerospace engineering, contributing to the advancement of reliable and efficient aircraft electrical systems.

## 5.3 Balance Between Model Accuracy and Simulation Speed

Most importantly, one has to find a model accuracy/simulation speed balance when creating and implementing the Simplified Synchronous Machine SI Units model using MATLAB/Simulink. It is possible through the latter to achieve realistic simulations with an assured computationally efficient simulation performance to give room for fast iterations and detailed analysis, but without compromising the reliability of those results. The fundamental trade-off/challenge in simulation modeling is that of accuracy versus speed: Models of high fidelity tend to be computationally heavy and require more time for simulation. On the other hand, simplified, computationally fast models hardly have the level of detail to correctly capture all the relevant system behaviors. The trick lies in deriving a model that works accurately enough to derive value, yet is running efficiently enough to be used in a common iterative design and test environment.

One of the element factors helping in achieving this balance in the model is the use of Simplified Synchronous Machine blocks, developed to implement the series RL impedance and voltage sources to represent the series and parallel compensating reactors, respectively, defusing the computational load yet ensuring the right representation of the electrical characteristics of the machine. Mainly considering the most important parameters—resistance, inductance, and mechanical dynamics (inertia constant, torque, damping factor, and rotor speed)—the model will maintain a high level of accuracy but without unnecessary complexity.

The other element is the selection of the simulation mode. The powergui block under MATLAB/Simulink allows operation in all sorts of modes—continuous, discrete, and phasor. For this work, the continuous mode is considered in the quest for detail and accuracy. However, in cases where speed is more important, the switching to discrete or phasor modes can reduce simulation time

by orders of magnitude and still provide sufficient accuracy for some analyses. These modes reduce the complexity of the system dynamics representation, and hence increase simulation speed.

The optimization tools of MATLAB include the Simulink Design Optimization and MATLAB Optimization Toolbox, and are employed to further fine-tune the model parameters. Sensitivity analysis allows the determination of the significance of the different parameters on the outcome of the simulation. By focusing on which parameters are most significant, an optimization of the model to obtain the best possible accuracy with the least computational effort can be achieved. This helps make the model robust and efficient enough to be relied upon for accurate results in a reasonable time frame.

Validation of the model against experimental measurements and recognized benchmark data is an essential aspect to guarantee the validity of the simplified model with respect to the real configuration. This iteration of model validation is achieved by matching the simulation results with the real test results; this is achieved by correcting the model. The performance of the model in terms of detail is monitored and constantly refined to maintain the balance between detail and efficiency. This process is iterative in the nature of the model-based design. The first few simulations give experience that can be used in the refinement of the model for the subsequent iterations, hence improving the accuracy and speed. This is a highly iterative approach that has as its end goal the relative balance of accuracy and speed within the various solution scenarios.

Practically, this means that it should be possible to use the developed model across a wide range of applications that include preliminary design and feasibility studies, through to detailed performance analysis and optimization. In simulation, the ability is really both very accurate and very fast for running the simulations, which serves as a process support to the critical decision-making for the development of aircraft electrical systems, hence an advanced support for aerospace engineering.

#### 5.4 Comparison with Previous Studies

A closer look into the simulation results gives a better insight into the performance and behavior of the Simplified Synchronous Machine SI Units model under different operational conditions. The simulation results are consistent with several previous works, affirming the model's reliability. For example, the overall behavior of the synchronous machines in our model agrees with the findings of Khan and Wang (2019), who demonstrated that detailed models incorporating series RL impedance and voltage sources can accurately represent the electrical characteristics of synchronous machines.

Furthermore, the stability of our model during transient events, such as sudden load changes, aligns with observations made by Patel and Singh (2020). They emphasized the importance of high-fidelity models in capturing dynamic behaviors under transient conditions. Our simulations confirmed that the model could quickly adjust to new operating points with minimal oscillations, reflecting similar performance stability reported in their research.

The validation process, involving comparisons of simulation results with experimental data, mirrors the approach taken by Smith et al. (2021). Their review highlighted the necessity of aligning model outputs with real-world measurements to ensure accuracy. Our findings showed a high degree of correlation with experimental data, supporting the model's reliability, much like the validated models discussed in their comprehensive review.

Additionally, the use of optimization tools for sensitivity analysis and parameter tuning in our study is in line with the techniques explored by Jones and Roberts (2023). Their research on model-based design optimization of aircraft electrical systems demonstrated that iterative optimization is crucial for balancing model detail and computational load. Our study similarly employed MATLAB's optimization tools, resulting in improved model accuracy and efficiency.

The advancements in simulation speed achieved through careful selection of simulation modes and optimization of critical parameters are comparable to the methods described by Miller et al. (2022). While our study primarily used continuous mode in the powergui block, the principles of balancing accuracy with simulation speed were applied similarly. Our results confirmed that efficient simulations could be achieved without sacrificing detail, reflecting advancements in simulation methodologies described in their work.

Moreover, the use of the Three-Phase Series RLC Load block to simulate balanced load conditions aligns with the methodologies outlined by Zhang et al. (2020). Their research on hybrid simulation techniques highlighted the importance of accurate load modeling to assess system performance comprehensively. Our study corroborated these findings, demonstrating that precise load representation is essential for realistic simulations.

In summary, the findings of this study are consistent with those of previous research, confirming the validity and reliability of the Simplified Synchronous Machine SI Units model in MATLAB/Simulink. The detailed comparison with existing literature underscores the model's capability to accurately simulate and analyze aircraft electrical systems, contributing to ongoing advancements in this field. This consistency with established research highlights the robustness of the methodologies employed and the potential for further applications and improvements in aerospace engineering.

## VI. OPTIMIZATION OF SYSTEM PERFORMANCE

### 6.1 Optimization Techniques

Optimization in the performance of the Simplified Synchronous Machine SI Units model at the system level involves a suite of techniques and tools to ensure proper execution of the model without loss of accuracy. A major technique within this is sensitivity analysis and parameter tuning—directly correlated with adjusting the model parameters for minimum errors and maximum performance. This is implemented using tools such as Simulink Design Optimization and MATLAB Optimization Toolbox, and realized through algorithms based on gradient descent optimization, genetic algorithms, and particle swarm optimization. These algorithms systematically explore the parameter space to find the optimal configurations.

Another important technique is sensitivity analysis, which involves assessing the effect of each parameter on the system's output by varying each parameter systematically. This focus on critical parameters enhances efficiency without unnecessary computational resource expenditure, ensuring robustness under various operating conditions.

Model reduction is also crucial in this form of optimization. Simplifying the model by reducing the number of states and parameters without compromising essential dynamics significantly improves simulation speed. Techniques like balanced truncation, proper orthogonal decomposition, and singular value decomposition create reduced-order models that retain accuracy but require less computational power. Balancing between model complexity and performance in real-time applications and iterative design processes becomes paramount.

## 6.2 Ensuring System Requirements

The availability of a rigorous validation and verification process will ensure that the model meets all the system requirements. This process starts with defining clear requirements based on the intended application, such as accuracy, stability, response time, and robustness under various operating conditions. The model is then verified against these detailed specifications to determine if the demands are met.

Verification involves checking if the model correctly implements the system's specifications, which can be done through formal verification methodologies, including model checking. Tools like Simulink Verification and Validation automate these checks and generate test cases for different scenarios, ensuring comprehensive verification.

Validation involves comparing the model's output with real-world data or benchmark results to ensure it accurately represents the physical system. This can include laboratory experiments, field tests, or comparisons with results from other validated models. Discrepancies are investigated to refine the model parameters and structure, ensuring it meets system requirements.

Robustness testing, including simulation methods like Monte Carlo simulations where the model is run multiple times with varying inputs, helps understand the impact of uncertainties and variations on performance. This ensures the model remains stable and reliable under extreme conditions, confirming its robustness.

## 6.3 Model Improvement Strategies

Model efficiency and reliability improvement is continuous and follows various strategies. Some important strategies include continuous integration of feedback from simulations and real applications. Importantly, with every update of new data and insights, the model remains up-to-date, and errors are corrected or new features are included (Jones & Roberts, 2023).

Another strategy supporting the above is modular design. This approach divides the model into smaller, manageable sub-models representing parts of the system. This effectively reduces the complexity of the whole system. Modular design allows for individual testing and optimization of each sub-model, which can then be integrated into the larger system. It also facilitates easier maintenance and updates, as changes in one module do not necessarily affect others (Patel & Singh, 2020).

Advanced numerical methods and solvers, such as adaptive time-stepping algorithms, implicit solvers, and parallel computing, reduce simulation time while maintaining accuracy. This ensures efficient use of computational resources, particularly for large-scale models requiring high precision (Miller et al., 2022).

Model efficiency can be further improved with machine learning techniques. Techniques such as reinforcement learning and neural networks create predictive models and optimize control strategies. Training these models with historical data provides insights and predictions that conventional methods might miss. Integrating machine learning with traditional modeling approaches offers even more accurate and efficient system techniques (Zhang et al., 2020).

To sum up, optimizing system performance involves a mix of parameter tuning, sensitivity analysis, and model reduction. Stringent verification, validation, and robustness testing ensure the model meets system requirements. This, combined with continuous improvement through feedback integration, modular design, advanced numerical methods, and machine learning techniques, guarantees that the model remains efficient and reliable over time. Together, these strategies contribute to the development of robust, accurate, and efficient models for aircraft electrical systems.

## VII. CONCLUSION

The research on the Simplified Synchronous Machine SI Units model in MATLAB/Simulink provides insight into the modeling, simulation, and optimization of aircraft electrical systems. Major results of the study posit that it presented a model that reflects the electrical and mechanical dynamics of synchronous machines in a manner that encourages realistic simulation under

different operational scenarios. Optimization of the parameter tuning, sensitivity analysis, and model reduction gave a model that is computationally efficient and provides a good trade-off between accuracy and computational complexity. Verification against experimental results and robust testing showed the model to be reliable and align well with previous studies in the field.

Future research should focus on a more advanced model with the integration of machine learning techniques applied in predictive maintenance and real-time optimization. The model should be subjected to studies with different types of loads and load configurations to test its stability and performance. Additionally, a deeper representation of other units in the aircraft electrical network will make the model a more complete tool for system analysis and design.

The applicative utility of the model in the domain of aircraft electrical network design is enormous. It is very useful at the initial design stages for testing different configurations and optimizing system parameters before actual physical installation. The model could also be useful for maintenance and troubleshooting, enabling engineers to simulate and diagnose potential problems in a controlled environment. Moreover, the insights developed from this model could be transferred to the design of electrical systems that will be more reliable and efficient, contributing to the advancement of aerospace engineering and technology.

## REFERENCES

- Diston, D.J. (1999). Unified Modelling of Aerospace Systems: A Bond Graph Approach. PhD thesis, University of Glasgow.
- Gandhi, N., & Yedavalli, R. K. (2015). Model-based design and simulation of aircraft electrical systems using MATLAB/Simulink. *Journal of Aerospace Engineering*, 28(2), 04014067.
- Gawthrop, P.J., & Bevan, G.P. (2007). Bond-graph modeling. *IEEE Control Systems Magazine*, 27(2), 24-45.
- Herber, D.R. (2018). Advancing Model-Based Engineering through Improved Integration of Domain-Specific Simulation and Analysis using SysML-based Models. PhD thesis, Colorado State University.
- Johnson, M. P. (2018). Modeling of synchronous machines in aircraft electrical networks using simplified SI units. Master's Thesis, Massachusetts Institute of Technology.
- Jones, M., & Roberts, L. (2023). Model-based design optimization of aircraft electrical systems using MATLAB/Simulink. *Aerospace Science and Technology*, 52, 45-55.
- Khan, A., & Wang, Y. (2019). Simulation of aircraft electrical power systems using MATLAB/Simulink and Simscape. *IEEE Transactions on Aerospace and Electronic Systems*, 55(3), 1124-1133.
- Koen, B.V. (2018). Application of Agile Model-Based Systems Engineering in aircraft conceptual design. *The Aeronautical Journal*.
- Kypuros, J. (2013). *System Dynamics and Control with Bond Graph Modeling*. CRC Press.
- Lee, J., & Kim, S. (2019). Advanced simulation techniques for aircraft electrical systems using MATLAB/Simulink. *Proceedings of the IEEE International Conference on Aerospace Electronics and Systems*, 467-472.
- MathWorks. (2024). Model-based systems engineering (MBSE) with MATLAB & Simulink. Retrieved from MathWorks
- MathWorks. (2024). Modeling, Simulation, and Flight Control Design of an Aircraft with Simulink. Retrieved from MathWorks.
- Miller, S., Brown, T., & Garcia, H. (2022). Enhancing aircraft electrical network modeling with hybrid simulation techniques. *IEEE Access*, 10, 56378-56388.
- Patel, R., & Singh, P. (2020). Detailed modeling of aircraft electrical networks with Simscape. *Journal of Aerospace Engineering*, 33(4), 123-134.
- Servotech Inc. (2024). Model-based design using MATLAB and Simulink. Retrieved from Servotech Inc
- Smith, J., Thompson, R., & Lee, D. (2021). A review of simulation studies using MATLAB/Simulink for aircraft electrical systems. *International Journal of Aerospace Engineering*, 45(2), 89-102.
- Zhang, Y., Li, X., & Wang, Q. (2020). Hybrid simulation techniques for modeling aircraft electrical networks. *Proceedings of the International Conference on Aerospace Electrical Systems*, 235-240.